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Problems associated with ionosphere study by using tiny satellite

Institute of Space and plasma Sciences, No.1 University Rd, Tainan ,70101, Taiwan /International Center for Space Weather Study and Education, Kyushu University E mail;oyama@pssc.ncku.edu.tw

Aim of the paper, and summary Tiny satellites ($\sim 10-20$ kg) might be able to provide excellent platforms in many science fields in the near future when it is used for the constellation. However the tiny satellite has its own problems, which should be overcome now to prepare for the near future mission. For ionosphere satellite, two serious problems appear when we try to use conventional DC Langmuir probe to measure Te and Ne.

. These are, (1) insufficient conductive surface area of satellite, (2) Contaminants attached on both electrode and satellite wall. We discuss these two problems and suggests possible countermoves: an instrument which is not influenced by contamination as well as that of surface area.

We divide our discussion into two parts.

 Problems which come out from tiny satellite , conductive surface area as counter electrode, and contamination of the electrode and satellite surface

2. One solution to the above

Principle of Electron Temperature probe



Problems of counter electrode



Fig. 2 The I-V curves for different satellite areas as surface area ratio of electrode and satellite surface becomes 1, ion and electron current is equal

In addition to surface area ,effect of contamination layer of satellite frame needs to be considered

1. When conductive surface area of the satellite(Asa) is much larger than the electrode surface area (Ap) (1000 times), I-V curve is simply controlled by electrode sheath.

2. When Asa is the same order of Apr, I-V curve becomes complicated . Capacitance effect of contamination of satellite wall can not be neglected. ...

For large probe current, fast sweep of probe bias reduces contamination effects. <u>While, for small satellite,</u> <u>the probe current is an order of ion current</u>, <u>which reduces frequency response of the pre amplifier</u>.

Serious effect of contamination !



Current –voltage characteristic of a conventional DC Langmuir probe (assuming infinite surface area of counter electrode)

Electrode



Fig.3 Electric current system of probe- plasma –satellite system :Small satellite surface area

What will happen for small surface area $(A_{sa}/A_p=19.48)$?



Fig.4 Hysteresis of I-V curve for small counter electrode

Can we have a solution to get accurate Te and Ne by small satellite ?- YeS

How?

TeNeP :modification of ETP (Electron temperature probe is not influenced by electrode contamination, and can work for tiny satellite. HINOTORI Sun Observer (Launched ;1981 Feb Mission Termination ;June 1982

ELECTRODE

ELECTRONICS

Electron Temperature Próbe (Hirao,

K. and K.- I.Oyama, J.Geomag

Geoelectr., 22, 239-402, 1970.

5 Earth orbiting Japanese satellites, Korean, Russia, Brazil satellites, and more than 60 sounding rockets including India, US ,Brazil, West Germany, Canada

300g, 25mmH,100mm Dia. With 120 mm circular electrode with

10mm gap

Impedance Probe (Ne) Oya,H.,T.Takahashi,and S.Watanabe,, J.Geomag.Geoelectr.,38,111-124,1986.



AOGS Small Satellite Payload Task Group: S3PTG

http://space.geocities.jp/s3ptg/members.html

Under AOGS (Asia Oceania, Geoscience Society) we have established a group named above in 2013. The purpose of this group is to prepare for the future small satellite constellation for ionosphere study. The data which come out from this mission will be used for space weather Study. We would like to use this data to study the coupling among lithosphere, atmosphere-and ionosphere, such as effects of large earthquake on the ionosphere. In order to increase drastically the output from satellite mission, a constellation of small satellite is essential. In order to materialize the satellite constellation, we propose to work together among the countries who are interested in. Specifically one tiny satellite might be provided from individual member country.

Prof. Koichiro Oyama (National Cheng Kung University, Taiwan)

- oyama@pssc ncku.edu.tw
 Dr. Wing-Huen Ip (National Central University, Taiwan)wingip@astro.ncu.edu.tw
 Prof. Kyoung Min (Korea Advanced Institute of Science and Technology, Korea, South) kwmin@kaist.ac.kr
 Mr. Ludmil Bankov (Space Research and Technology Institute, Bulgaria)ludmil.bankov@gmail.com
 Dr. Borra Reddy (CSIR, India) borramreddy@rediffmail.com
 Prof. Minakshi Devi (Gauhati University, India) md555gu@gmail.com
 Dr. Dhiren Kataria (Mullard Space Science Laboratory, United Kingdom),dokataria@gmail.com
- Prof. J. Y. Liu (National Central University),
- For detail information ,please contact
- To Mr. Tetsuya Kodama (, Task Group secretariat ,Japan Aerospace Exploration Agency, Japan) kodama.tetsuya@jaxa.jp

Concluding Remarks

Two serious problems should be taken into account for tiny satellite mission to measure Te and Ne of ionosphere plasma: Insufficient conductive surface area of satellite, and contaminants attached on both electrode and satellite wall. To remove these problems we have proposed one instrument. We introduced Small Satellite Payload Task group which tries to prepare for near future mission.

Enjoy your stay at Fukuoka !!!

Slides following are appendix

Local time and geomagnetic latitude variation of Electron temperature at 600 km 600kmにおける電子温度地方時変化



電子温度は地方時、太陽活動度、緯度、経度、季節により変化する。

Su, Y. Z., K.-I. Oyama, G. J. Bailey, T. Takahashi and S. Watanabe, J. Geophys. Res., 100, 14591, 1995.

.Oyama, K.-I., M. A. Abdu, N. Balan, G. J. Bailey, S. Watanabe, T. Takahashi, E. R. de Paula, I. S. Batista, H. Oya and F. Isoda, J. Geophys.Res.,102,417, 1997.



observation respectively



M=7.4 M=6.6 техадоттлиботта Deviation of Te from the model (Averaged) versus days. For larger earth quake precursor and recovery become longer

Longitude-latitude Map of ΔTe for three EQs



دەەھەرەت ! Can not identify Epicenter ! Day is



Atomic ion density along satellite pass from 5 days prior to EQ_{MBSBRRFT}





Location of the minimum of $[O^+]_{REMOVED}$ (Upper panel), and reduction of $[O^+]_{REMOVED}$

2. When is D day HERE ?





Diversity of Te in the plasma bubble.oyama, K.-I., K. Schlegel, and S.

Watanabe, Temperature structure of plasma bubbles in the low latitude ionosphere around 600 km altitude, Planet. Space Sci., 36, No.6, 553-567, 1988.